

Name of the Student: \_\_\_\_\_

Max. Marks : 22 Marks

Time : 22 Minutes

Mark Schemes

**Q1.**

C

[1]

**Q2.**

(a) Equatorial orbit ✓

Moving west to east ✓

Period 24 hours ✓

**ANY TWO**

2

$$(b) \quad T \left( = \frac{2\pi}{\omega} = \frac{2\pi}{2.5(4) \times 10^{-4}} \right) = 2.5 \times 10^4 \text{ s } \checkmark$$

1

$$(c) \quad \lambda \left( = \frac{c}{f} = \frac{3.0 \times 10^8}{1100 \times 10^6} \right) = 0.27(3) \text{ m } \checkmark$$

$$\theta \left( = \frac{\lambda}{d} = \frac{0.27(3)}{1.7} \right) = 0.16(1) \text{ rad} = 92^\circ \checkmark$$

$$(\text{linear}) \text{ width} = D\theta = 12000 \text{ km } 0.16(1) \text{ rad} = 1.9(3) \times 10^3 \text{ km } \checkmark$$

3

(d) Angle subtended by beam at Earth's centre

$$= \text{beam width} / \text{Earth's radius} = 1.9(3) \times 10^3 / 6400 \text{ ) } \checkmark$$

0.30 rad (or 17°) ✓

$$\text{Time taken} = \alpha / \omega = 0.30 / 2.5(4) \times 10^{-4} = 1.18 \times 10^3 \text{ s}$$

= 20 mins ✓

*Alternative:*

$$\text{Speed of point on surface directly below satellite} = \omega R$$

$$= 2.5(4) \times 10^{-4} \times 6400 \times 10^3 \text{ )}$$

$$= 1.63 \times 10^3 \text{ m s}^{-1} \checkmark$$

$$\text{Time taken} = \text{width} / \text{speed}$$

$$= 1.93 \times 10^6 \text{ m} / 1.63 \times 10^3 \text{ m s}^{-1} \checkmark$$

$$= 1.18 \times 10^3 \text{ s}$$

(accept  $1.2 \times 10^3 \text{ s}$  or 20 mins) ✓

or

Satellite has to move through angle of  $1900 / 6400$  radian = 0.29 rad ✓

$$\text{Fraction of one orbit} = 0.30 / 2 \times 3.14 \text{ ✓}$$

$$\text{Time} = 0.048 \times 2.5 \times 10^4 = 1.19 \times 10^3 \text{ s ✓}$$

$$\text{Time} = \frac{17}{360} \times 2.5 \times 10^4 = 1.18 \times 10^3 \text{ s}$$

or

$$\text{Circumference of Earth} = 2\pi \times 6370 \text{ ✓}$$

$$= 40023 \text{ km}$$

$$\text{Width of beam at surface} = 1920 \text{ km ✓}$$

$$\text{Time} = \frac{1920}{40023} \times 2.48 \times 10^4$$

$$= 1180 \text{ s} = 19.6 \text{ min ✓}$$

3

(e) Signal would be weaker ✓ (as distance it travels is greater)

Energy spread over wider area/intensity decreases with increase of distance ✓

Signal received for longer (each orbit) ✓

Beam width increases with satellite height/satellite moves at lower angular speed ✓)

4

[13]

### Q3.

(a) (i)  $M = \frac{4}{3} \pi R^3 \rho \text{ ✓}$

combined with  $g_s = \frac{GM}{R^2}$  (gives  $g_s = \frac{4}{3} \pi G R \rho$ ) ✓

Do not allow  $r$  instead of  $R$  in final answer but condone in early stages of working.

Evidence of combination, eg cancelling  $R^2$  required for second mark.

2

(ii)  $R = \left( \frac{3g_s}{4\pi G\rho} \right) = \frac{3 \times 8.87}{4\pi 6.67 \times 10^{-11} \times 5.24 \times 10^3} \text{ ✓}$

gives  $R = 6.06 \times 10^6 \text{ (m) ✓}$

answer to **3SF** ✓

SF mark is independent but may only be awarded after some working is presented.

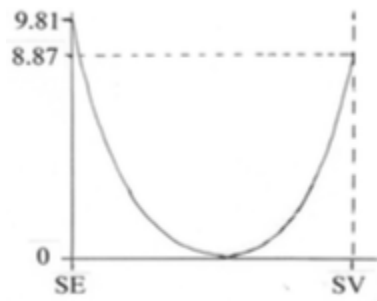
3

(b) line starts at 9.81 and ends at 8.87 ✓

correct shape curve which falls and rises ✓

falls to zero value near centre of and to right of centre of distance scale ✓

[Minimum of graph in 3rd point to be  $>0.5$  and  $<0.75$  SE-SV distance]



*For 3rd mark accept flatter curve than the above in central region.*

3

[8]